

### SPECIFICATION AND DESCRIPTION

### **P2002 SIERRA MKII**

This document applies only to the Tecnam P2002 SIERRA MKII and is published for the purpose of providing general information for the evaluation of design, powerplant, performance and equipment.



Should more information be required, please contact:

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# GENERAL DESCRIPTION P2002 SIERRA MkII

#### **Next-generation Light Sport Aircraft**

#### Construction

The P2002 Sierra MkII is a two-seater side by side, low wing aircraft. The Sierra MkII features superlative performance and flying qualities, confirmed by hundreds of P2002 aircraft sold throughout the world. The ease of piloting and maintenance make this aircraft an excellent solution for training in fight schools. It is also an ideal platform for surveillance and, of course, for pure recreational and private use. The option to use 100LL AVGas or unleaded automotive fuel (with up to 10% ethanol content) makes this aircraft even more flexible and cost effective.

The P2002 Sierra MkII encompasses the latest developments in Tecnam aircraft. The use of advanced software for design, structural and fluidynamics analysis, and experience in building airplanes using all types of materials, results in continuous aircraft improvement. Due to the tapered laminar airfoil and the slotted flaps the P2002 Sierra MkII is an outstanding aircraft with the perfect mix of aerodynamics, performance, and structural efficiency.

Many flight schools in Europe and all over the world

rely on the P2002 for students' initial training. Many of them continue their training up to the ATPL using the Tecnam P2006T twin, making Tecnam the ideal one-stop-shop for training aircraft all over the world. The Tecnam P2002 Sierra MkII's structure is based on a steel tube cabin truss covered on the forward fuselage by carbon fiber panels while the tail cone is covered by light alloy sheets. The wing is all light alloy built with a single spar and full metal torsion box. The wing's leading edges are easily detachable for repairs and also incorporate the fuel tanks (110 L - 29 US Gal in total). They are separated from the cabin in order to maximize passive protection. The sliding canopy allows 360° of vision in the cockpit and has full rollover protection tested via inverted drop tests.

The horizontal stabilator tail design, provides remarkable longitudinal hands-off stability along with minimum drag and weight penalty. This provides balanced two finger flight control. The wide slotted flaps, are electrically activated and allow stall speed lower than 40 Kts, and allow the aircraft to perform steep approaches and easier landings. The all alloy ailerons are effective and ensure a quick roll rate without being overly sensitive. All control surfaces are made out of light

alloy and all, except for flaps and tab, are massbalanced.

#### **Landing Gear**

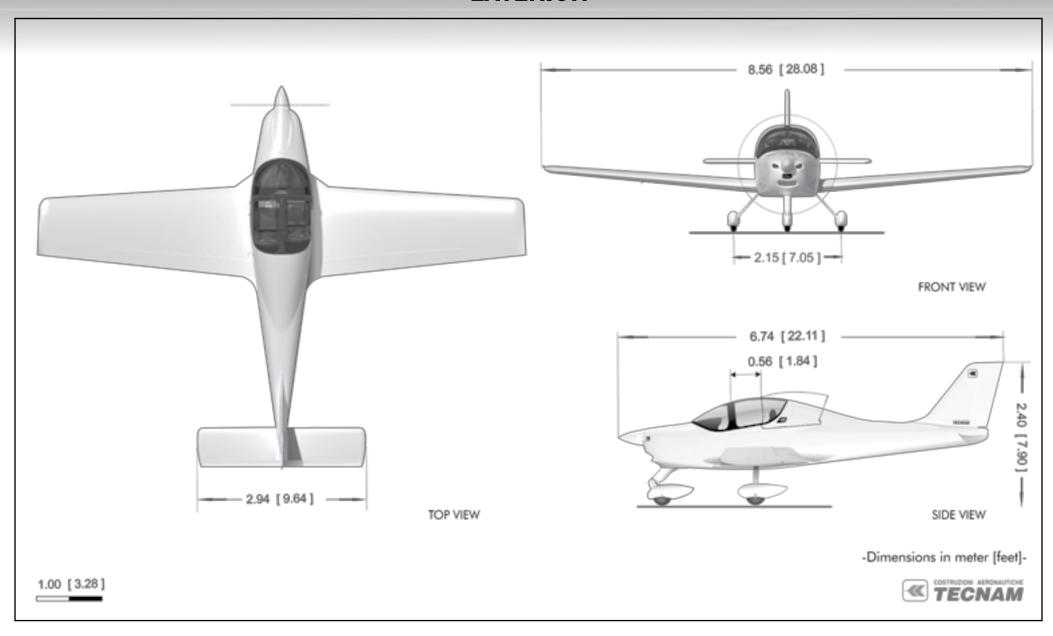
The main landing gear is composed of light alloy spring. This provides a main gear that is robust for unimproved landing strips and requires no service. The trailing link nose gear uses a rubber shock absorber system that was designed for the rigours of the training environment. The main landing gear wheels and brakes are 5.00x5, providing the facility to use multiple different tire brands that can be chosen in relation to the mission type and expected number of landings per hour. There are toe brakes, with a parking brake valve located on the console between the seats.

#### Certification

This model is available as Advanced Ultralight & US/LSA.

The Tecnam P2002 Sierra MkII is ready for the new EU 600kg ULM category. This model is available in those Countries where this category is approved by their Local Civil Aviation Authorities.

### **EXTERIOR**



### **DIMENSIONS**

Wing	ft	m
Span (overall)	28.08	8.56
Area	123.8ft <sup>2</sup>	11,5 mq
Dihedral	5°	
Aspect ratio		6.4

Dimensions	ft	m
Overall Height	7.9	2,4
Overall Length	22.1	6,74

WEIGHT AND LOADING	SIERRA MKII	
WEIGHT AND LOADING	kg	lb
Maximum Take-Off Weight	600	1,320
Empty Weight, Standard	367	809
Useful Load	233	514
Baggage allowance	20	44

DEDECOMANCE			
PERFORMANCE	912 ULS2 600 kg		
Max Cruise Speed KTAS	122 kts	226 km/h	
Stall Speed (Flaps Down Power Off) KCAS	40 kts	74 km/h	
Practical ceiling	14000 ft	4267 m	
Take off run	715 ft	218 m	
Take off distance	1.115 ft	340 m	
Landing Run	505 ft	154 m	
Landing Distance	991 ft	302 m	
Rate of climb	923 ft/min	4,7 m/sec	
Range	620 NM	1148 km	



### **POWERPLANT & ACCESSORIES**

The top and bottom engine cowls are quickly and easily removable making any maintenance procedure faster to accomplish. The top cowl has two large hinged gull-wing style doors for easy access and effective preflight inspection of the entire engine compartment. The engine is set low and the cowling slopes down from the windshield, so forward visibility is outstanding even with a fully equipped instrument panel. The steel firewall is sound proofed and the power plant is a liquid and air cooled Rotax 912ULS2 four- cylinder, four- stroke engine with an integrated 1:2.4286 reduction gear.

The use of liquid cooled heads and air cooled cylinders allows the engine to maintain safe operating temperatures even if a rapid descent is performed immediately after a prolonged climb. A fixed pitch wooden/composite wrapped Sensenich propeller comes as standard whit a ground adjustable pitch propeller; a constant speed V.P. propeller are available as options. The quick drain gascolator is installed under the cabin door and provides easy access for checking fuel. An electric fuel pump is installed to provide an effective back- up to the mechanical one. The battery is located in the rear of the fuselage with easy access through an external hinged door.

### **ROTAX 912 ULS2**



- 4-cylinder
- 4-stroke liquid-/air-cooled engine with opposed cylinders
- Dry sump forced lubrication with separate oil tank, automatic adjustment by hydraulic valve tappet
- · Mechanical fuel pump
- · Dual electronic ignition
- · Propeller speed reduction unit
- · Gearbox Reduction Ratio 2.43:1



### STANDARD EQUIPMENT

### Flight instruments and indicators

Magnetic Compass

Airspeed Ind.,

Altimeter (In/Mb)

Vertical Speed

Bank Indicator

Pitot System

Static System

#### **Engine Instruments & Indicators**

Head Temperature, CT

Voltmeter

Oil Temperature

Oil Pressure

Fuel Pressure

**RPM** 

Trim

Flaps

LH+RH Fuel qty

#### Flight controls

Hydraulic Toe Brakes

Parking Brake

Electrical Flaps, Preselect

**Dual Flight Controls** 

Steerable Nose Wheel

Engine Controls:

- Central Quadrant With Single Throttle Lever
- Choke

Stabilator Trim, Electric (Controlled From Stick)

Fuel Control Selector With LH/RH Off (Andair)

#### **Electrical system**

12 VOLT 18A AMP. Battery, Alliant 12 VOLT Alternators 20 AMP.

Starter Key ACS

Split Starter

**Rocker Switches:** 

- Avionic Master
- Fuel Pump
- Landing light

Circuit Breaker Panel

12 Volt socket

Warning Lights:

- Alt Out

#### **Fuel system**

Two Integral Fuel Tanks With 110 Litres 29 US Gal Total Capacity Engine Driven Fuel Pump Auxiliary Fuel Pump, Electric Fuel Tank Quick Drain, Two

1 X Shut Off Valve

#### **Powerplant and Propeller**

Firewall

Engine 1 Rotax 912ULS2 100 Hp

4 Cylinders

Liquid/Air Cooled

Integrated Reduction Gear

**Dual Ignition System** 

Throttle Control (Central)

**Tubular Steel Engine Mount** 

Propeller: Sensenich 2 Blade Fix Pitch

Propeller Spinner

Two Air Filter

Oil Filter

Oil And Water Coolers

Fire Sleeve Fuel And Oil Tubes

#### Exterior

Epoxy Corrosion Proofing, All Structure

Canopy, Lock And Key

Tie Down Rings

Main Wheels, 5.00 X 5 - Nose 5,00 X 5

#### **Exteriors lights**

Landing Light

#### Interior

Pilot And Co-Pilot Seats simulated leather adjustable fore and aft

Seat Belts 4 Points All Seats

Wall To Wall Carpeting
Map & Storage Pockets

Radio Call Plate

Soundproofing

**Luggage Compartments** 

**Emergency Hammer** 

#### Cabin comfort system

Windshield Defroster

Ventilator Adjustable

Heating System

#### **Product Support and Documents**

Manufacturer's Full Two Year Limited Warranty

Pilot's Operation Handbook

Maintenance Manual

Parts Catalogue

### **ANALOGUE**

EU 600 Kg Package for analogue





S-1002 P2002 SIERRA MKII ROTAX 100 HP EU 600 KG

Also Includes the following equipment:

#### Flight instruments and indicators

- GARMIN GMA245 Audio Panel
- GARMIN GTR225A COM 8.33 Khz with Antenna and Inst.
- GARMIN GTX 335 TRANSPONDER with GPS
- KANNAD ELT 406 Mhz WITH REMOTE MOUNTED SWITCH
- STICK PUSH-TO-TALK SWITCH-PILOT/COPILOT
- MIC & PHONE JACKS-PILOT/COPILOT

#### Antennas:

- Transponder
- VHF
- ELT
- Emergency Parachute System BRS

### **GLASS G3X**



### EU600KG category



Non-Additive. Replaces all Standard Equipment. - Also Includes the following equipment:

- Emergency Parachute System BRS

#### S-1003 P2002 SIERRA MKII 100 HP - G3X Two Screens

Non-Additive. Replaces all Standard Equipment. - Also Includes the following equipment:

#### Two Display GDU 460 With EIS

- · GDU460 PFD
- GDU460
- ADAHRS GSU 25
- GEA 24 Engine Instrument Module
- · GMU 22 Magnetometer
- GTP 59 Temperature Probe
- LRU KIT
- Installation Kit
- G3X Sensor Kit
- · GA56 Antenna

#### **GSU Configuration**

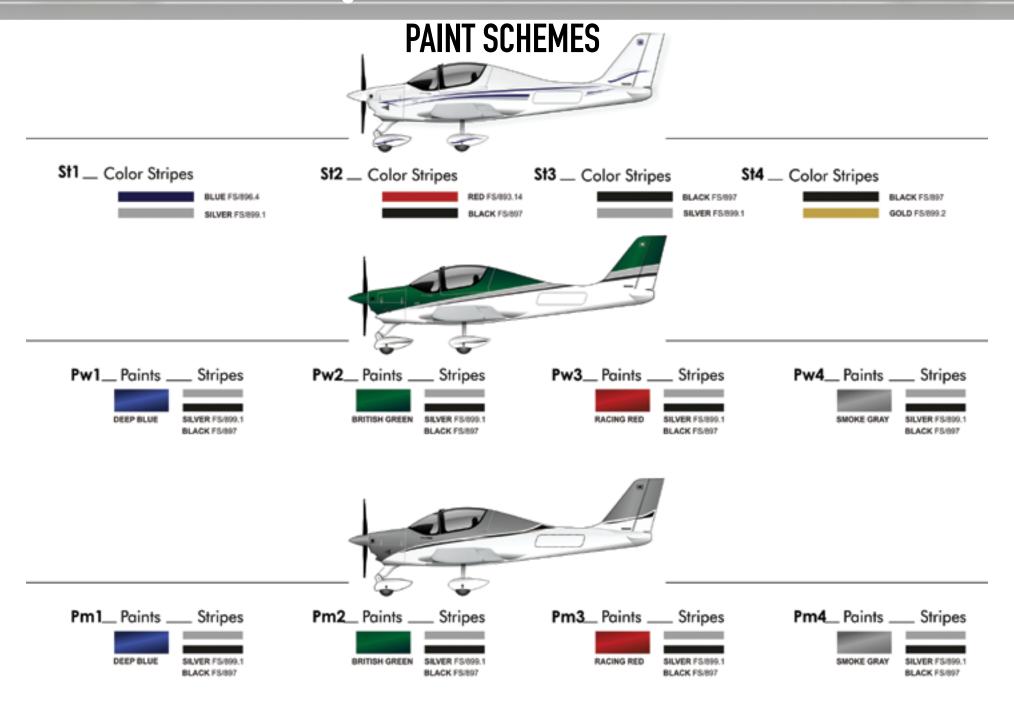
- · Amps (Ammeter Shunt Or Hall Effect)
- Monitor CHT
- Aircraft Bus Voltages
- Oil Temperature
- OAT
- · Oil Pressure
- · Manifold Pressure
- · Fuel Pressure
- RPM
- · Trim Indicator
- · Flaps Indicator
- · LH + RH Fuel Qty

#### **Avionics Package**

- · Garmin GMA 245 Audio Panel
- Garmin GTR 225A Com
- GTX 35R Transponder Mode S Remote Mounted
- · KANNAD ELT 406 Mhz With Remote Mounted Switch
- Stick Push-To-Talk Switch-Pilot/Copilot
- · Mic & Phone Jacks-Pilot/Copilot

#### Antennas:

- Transponder
- VHF
- GPS
- ELT



## P2002 SIERRA MKII EU-600Kg

NOTES



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